

**University of Technology**  
**Department of Laser & Opto-electronic Engineering**  
**Final Examination 2011-2012**

**Subject: Thermo. & fluid mechanics**  
**Division: Laser Engineering**  
**Examiner: Assist. Prof Khalid Salem**

**Class: 2<sup>nd</sup> year**  
**Time: 3 hours**  
**Date: 5/6/2012**



**Answer five questions only**

Q1) A) One kg of a certain gas is at  $0.11 \text{ MN/m}^2$  &  $15^\circ \text{C}$ . It is compressed isothermally until its volume is  $0.1 \text{ m}^3$ . Calculate the final pressure and temperature. Calculate also work done, change in internal energy and heat transfer, assume  $C_v = 0.6 \text{ kJ/kg.K}$   $C_p = 0.94 \text{ kJ/kg.K}$ . (6 marks)

B) A gas whose pressure, volume and temperature are  $275 \text{ kN/m}^2$ ,  $0.09 \text{ m}^3$  and  $185^\circ \text{C}$ , respectively, has its state changed at constant pressure until its temperature becomes  $15^\circ \text{C}$ . How much heat is transferred from the gas and how much work is done on the gas during the process also obtain the difference in specific internal energy take  $R = 0.29 \text{ kJ/kg.K}$ ,  $C_p = 1.005 \text{ kJ/kg.K}$  (6 marks)

Q2) In a steady flow system, a substance flows at a rate of  $4 \text{ kg/sec}$ , it enters with a velocity of  $300 \text{ m/sec}$ , a specific internal energy  $2100 \text{ kJ/kg}$ , pressure of  $620 \text{ kN/m}^2$  and specific volume of  $0.37 \text{ kg/m}^3$ . At the exit the velocity is  $150 \text{ m/sec}$ , specific internal energy  $1500 \text{ kJ/kg}$ , pressure of  $130 \text{ kN/m}^2$  and specific volume of  $1.2 \text{ kg/m}^3$ . The fluid has a heat loss to the surrounding of  $30 \text{ kJ/kg}$  as it passes through the turbine. The difference in height between inlet and outlet is  $10 \text{ meters}$ ; determine the power developed by the turbine in MW. (12 marks)

Q3) The volumetric analysis of a gas is as follows:  $\text{H}_2$  60%,  $\text{N}_2$  20%,  $\text{O}_2$  20%. This gas is mixed with air in proportions 1 volume of gas to 2 volume of air. Take the molar specific heat capacities at constant volume for diatomic gas  $= 20.9 \text{ kJ/kg mol K}$ . Assume that air contains 21% oxygen by volume and the remainder is  $\text{N}_2$ . If the air-gas mixture is kept at  $25^\circ \text{C}$  and  $150 \text{ KPa}$  for temperature and pressure respectively. Determine for the air-gas mixture:

- a) the mean relative molecular mass of the mixture ( $M_{av}$ ) (2 marks)
- b) The value of the adiabatic index of the mixture ( $\gamma$ ). (2 marks)
- c) The characteristic gas constant for the mixture ( $R$ ). (2 marks)
- d) The density of air. (2 marks)
- e) The partial pressure and mass fraction of  $\text{N}_2$  in the mixture. (4 marks)

Q4) For the velocity meter shown in fig1, derive an equation that relate  $R$  and  $V_1$ . If  $R = 2 \text{ cm}$  and the diameter of the pipe  $D_1 = 20 \text{ cm}$ , what is the volume flow rate in  $\text{m}^3/\text{s}$ . (12 marks)

Q5) For a large tanks shown in fig2, what is the value of the distance  $R$ . (12 marks)

Q6) A) Change the polytropic index ( $n$ ) in the equation  $PV^n = \text{Constant}$ , from zero to  $\infty$ , plot  $P$ - $V$  diagram, show the possible processes. (6 marks)

B) Prove that  $C_v = \frac{R}{\gamma - 1}$ , where  $\gamma$  - adiabatic index,  $R$  - characteristic gas constant,  $C_v$  - specific heat at constant

volume.

(6 marks)

Fig 1

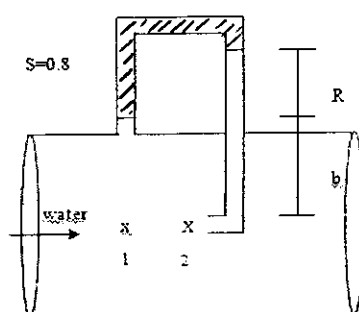
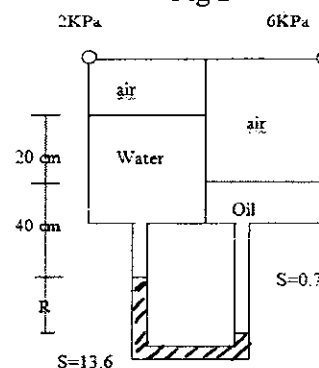


Fig 2

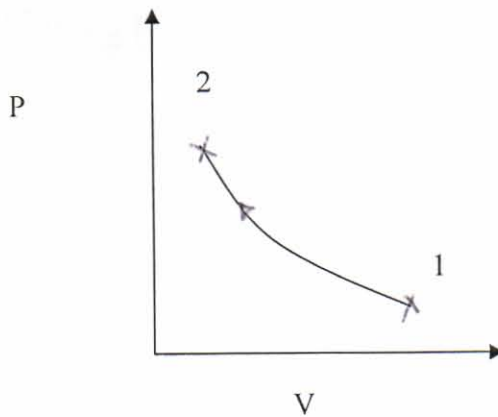


423-2-1

أجوبة أسئلة الترمو + الموائع  
المرحلة الثانية - الجزء 1  
2011/2012 دور اول

Q1)

A)



$$R = C_p - C_v = 0.94 - 0.6 \\ = 0.34 \text{ kJ/kg.K}$$

$$P_2 = \frac{mRT_2}{V_2} = \frac{1 \times 0.34 \times 1000 \times (273 + 15)}{0.1} = 979200 \text{ N/m}^2$$

$$T_2 = T_1 = 288^\circ \text{K}$$

$$w = PV \ln \frac{V_2}{V_1} = mRT \ln \frac{P_1}{P_2} = 1 \times 0.34 \times 1000 \times 288 \ln \frac{0.11}{0.9792} = -214078 \text{ J}$$

$Q = W$  since  $\Delta U = 0$  in isothermal process

$$\text{B) } m = \frac{P_1 V_1}{RT_1} = \frac{275000 \times 0.09}{290(273 + 185)} = 0.186 \text{ kg}$$

$$Q = mC_p(T_2 - T_1) \\ = 0.186 \times 1005(288 - 458) \\ = -31780 \text{ J}$$

$$V_2 = \frac{mRT_2}{P_2} = 0.0566 \text{ m}^3$$

$$W = p(V_2 - V_1) = 275000(0.0566 - 0.09) = -9190 \text{ J}$$

$$\Delta u = q - w$$

$$= \frac{-31780}{0.186} + \frac{9190}{0.186} = -121451.61 \frac{J}{kg}$$

Q2)

The steady state flow energy equation is

$$q + u_1 + P_1 v_1 + \frac{V_1^2}{2} + gz_1 = w + u_2 + P_2 v_2 + \frac{V_2^2}{2} + gz_2$$

$$-30000 + 2100 \times 1000 + 620 \times 1000 \times 0.37 + \frac{300^2}{2} + 10 \times 10 = w + 1500 \times 1000 + 130 \times 1000 \times 1.2 + \frac{150^2}{2}$$

$$w = 676750 J/kg$$

But

$$Power = \dot{m}w$$

$$= 4 \times 676750 = 2707000 \text{ Watts} = 2.707 \text{ MW}$$

Q3)

$$M_{av} = \frac{1(0.2 \times 28 + 0.6 \times 2 + 0.2 \times 32) + 2 \times (0.79 \times 28 + 0.21 \times 32)}{3} = 23.62 \text{ kg/kg.mol}$$

$$\bar{C}_{vav} = \frac{\sum n \bar{C}_v}{\sum n} = \left( \frac{1 \times (0.3 + 0.6 + 0.1) + 2 \times (0.79 + 0.21)}{3} \right) \times 20 = 20 \frac{kJ}{kg.mol.K}$$

$$R_m = \bar{C}_p - \bar{C}_v$$

$$\bar{C}_{pav} = 8.3143 + 20 = 28.3143 \frac{kJ}{kg.mol.K}$$

$$\gamma = \frac{\bar{C}_{pav}}{\bar{C}_{vav}} = \frac{28.3143}{20} = 1.4157$$

$$R_{av} = \frac{R_m}{M_{av}} = \frac{8.3143}{23.62} = 0.351 \frac{kJ}{kg.K}$$

$$P_{air} = \frac{P_t}{n_T} n_{air} = \frac{150}{3} \times 2 = 100 \frac{KN}{m^2}$$

$$R_{air} = \frac{R_m}{M_{air}} = \frac{8.3143}{(0.21 \times 32 + 0.79 \times 28)} = \frac{8.3143}{28.84} = 0.28829 \frac{kJ}{kg.K}$$

$$\rho_{air} = \frac{P_{air}}{R_{air} T} = \frac{100 \times 1000}{288.29(273 + 25)} = 1.164 \frac{kg}{m^3}$$

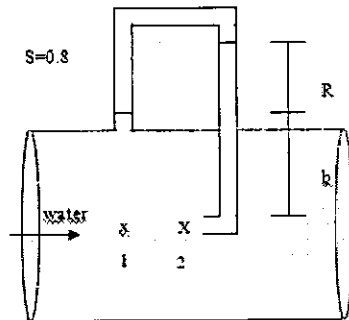
$$P_{N_2} = \frac{P_t}{n_T} n_{N_2} = \frac{150}{3} (0.2 + 2 \times 0.79) = 89 \frac{KN}{m^2}$$

$$m_T = n_T M_{av} = 3 \times 23.62 = 70.8 \text{ kg}$$

$$m_{N_2} = 28(0.2 + 2 \times 0.79) = 49.84 \text{ kg}$$

$$m_{N_2} \% = \frac{m_{N_2}}{m_T} = 70.5\%$$

Q4)



Bernoulli equation between 1&2, neglect loss then

$$\frac{P_1}{\gamma_w} + \frac{V_1^2}{2g} + z_1 = \frac{P_2}{\gamma_w} + \frac{V_2^2}{2g} + z_2 \quad \text{eq(1)}$$

$z_1 = z_2$  and  $V_2 = 0$  then eq(1) becomes

$$\frac{P_2 - P_1}{\gamma_w} = \frac{V_1^2}{2g} \quad \text{eq(2)}$$

Pressure equation 1&2

$$P_1 - b\gamma_w - RS\gamma_w + (R+b)\gamma_w = P_2 \quad \text{eq(3)}$$

Then

$$\frac{P_2 - P_1}{\gamma_w} = R(1 - S) \quad \text{eq(4)}$$

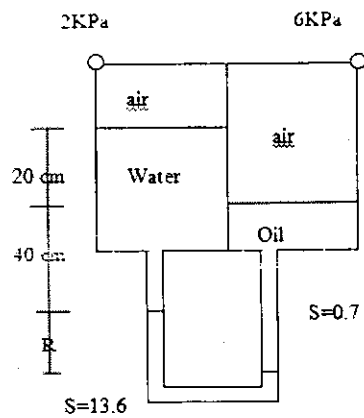
eq(2)=eq(4) then

$$V_1 = \sqrt{2gR(1 - S)}$$

$$V_1 = \sqrt{2 \times 9.8 \times 0.02(0.2)} = 0.28 \text{ m/s}$$

$$Q = V_1 A = 7.027 \left( \pi \frac{D^2}{4} \right) = 0.28 \left( \pi \frac{0.2^2}{4} \right) = 0.0088 \text{ m}^3/\text{s}$$

Q5)

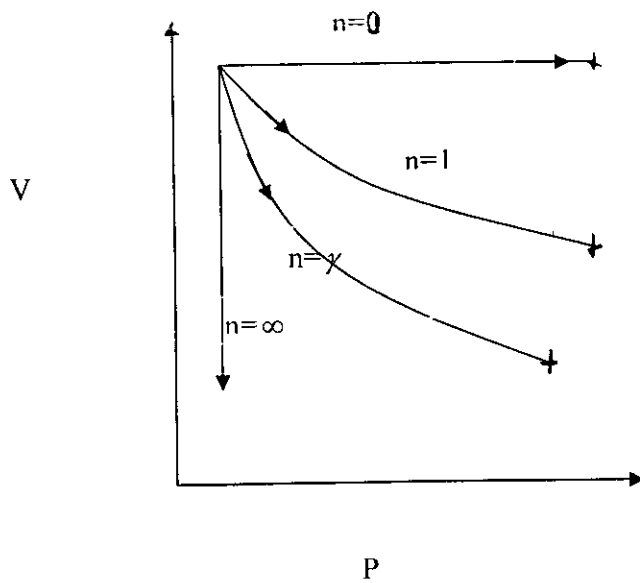


$$2000 + 0.6\gamma_w + 13.6R\gamma_w - 0.7\gamma_w(R + 0.4) = 6000$$

$$(13.6 - 0.7)R\gamma_w = 6000 - 2000 - 0.6\gamma_w + 0.7 \times 0.4\gamma_w$$

$$R = \frac{800}{12.9 \times 10000} = 0.0062 \text{ m}$$

Q6)



- $n=0$  constant pressure process
- $n=1$  isothermal process
- $n=\gamma$  adiabatic process
- $n=\infty$  constant volume process

B) Prove that  $C_v = \frac{R}{\gamma - 1}$ , where  $\gamma$ -adiabatic index, R- characteristic gas constant,  $C_v$ -specific heat at constant volume.

$$R = C_p - C_v$$

$$\gamma = \frac{C_p}{C_v}$$

then

$$\frac{R}{C_v} = \gamma - 1$$

$$C_v = \frac{R}{\gamma - 1}$$